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FINAL REPORT THERMAL PROPERTIES OF

TUNGSTEN-URANIUM DIOXIDE MIXTURES

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ABSTRACT

The thermal expansion, density, thermal diffusivity, and specific heat of various tungsten-uranium dioxide composites were measured from room temperature to very high temperatures. Because of the sample geometry, it was necessary to account for the finite pulse time effects in the thermal diffusivity measurements. The mathematical expression for a square wave energy input was derived and experimentally verified. Thermal conductivity values were calculated from the density, specific heat, and diffusivity results.

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THERMAL PROPERTIES OF TUNGSTEN-URANIUM DIOXIDE MIXTURES

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SUMMARY

The objective of this program was to determine the thermal expansion and thermal conductivity, from room temperature to 5000°F, of four composites (viz: 10, 20, 30, and 40 vol %) of uranium dioxide in a tungsten matrix.

The thermal expansions of the four composites were measured over this temperature interval, and equations for the results are given. The agreement between measured and calculated values is generally within 5%; but in a few cases, is as high as 8%. The expansions of these mixtures are, as expected, very close to that of pure tungsten.

The thermal conductivity was not measured directly. Instead, the diffusivities of the various mixtures were measured, and the thermal conductivities were calculated as the product of the specific heat, density, and diffusivity. The density and diffusivity were not corrected for thermal expansion. The expansion increases the diffusivity by the square of the increase in length of the sample, and decreases the density by the cube of this quantity. Consequently, the net effect is a 2% error in the thermal conductivity values at high temperatures. Since this is well within the ±9% uncertainty in the conductivity, it was considered unnecessary to correct the density and diffusivity values.

Conductivity values were also calculated using the Bruggeman variable-dispersion equation. These results are higher than the experimental values obtained from this study, partly because of anisotropic effects. In addition, the thermal conductivity of tungsten is not known accurately, and this may be the cause of some of the discrepancy.

The electrical resistivity of the 80 W - 20 UO₂ mixture was measured, and the results were used to calculate the contribution of free electrons to the total conductivity. It was found that the major portion of the heat transport is due to electrons.

I. INTRODUCTION

Space mission analyses indicate that the thrust requirements of some space exploration vehicles can be met through the use of nuclear propulsion systems. Advanced concepts of nuclear rockets require high operating temperatures for higher thrust and lower engine weights. A thermal nuclear rocket concept is presently being studied at the Lewis Research Center of NASA. This concept is based on the use of tungsten-uranium dioxide (W-UO₂) composites for the fuel element materials. These composites are currently being fabricated in the form of plates which are useful in initial studies to determine the behavior of this combination of materials at elevated temperatures.

To aid preliminary design analysis of this reactor concept, a knowledge of the properties of W-UO₂ composites is desirable. For this purpose, high-temperature mechanical properties of various W-UO₂ composites are being determined at the Lewis Research Center. Certain physical properties of various W-UO₂ composites have been studied at Atomics International, under NASA contract NAS3-4280. The specific objective of this study was to determine the thermal expansion and thermal conductivity values from room temperature to 5000°F for four compositions (viz., 10, 20, 30, and 40 vol %)* of uranium dioxide dispersed in a tungsten matrix. Since these values are highly dependent on the size and nature of the dispersed particles and on the fabrication history of the composites, the values determined in this program are to be considered only as indicative of the properties of the materials to be used in the actual fuel elements.

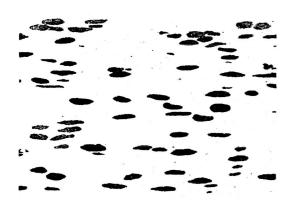
The present report constitutes the final technical summary of the experimental work performed in this physical property evaluation program. In this study, the thermal expansivities of four W-UO₂ composites were determined directly, by measuring the relative linear displacements of plate-type specimens as a function of temperature. Although thermal conductivities can also be measured directly, the sample configuration was not amenable to this type of measurement. Therefore, to simplify the thermal conductivity determinations, this property

^{*}In this report, percentages are expressed as volume percent (vol%), and tungsten is listed first.

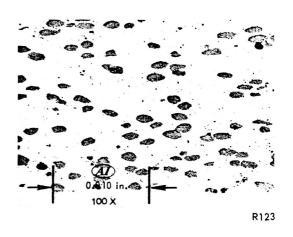
was obtained by measuring the thermal diffusivity (α), and calculating the thermal conductivity (k) from the relation:

$$k = \alpha c_{pd}$$
, ...(1)

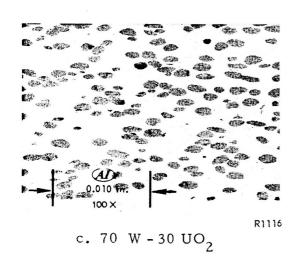
where c is the specific heat and d is the density. Since this method is as reliable as measuring the thermal conductivity directly, the former method was chosen for this program. This technique required measurement of density, thermal diffusivity, and specific heat. The results of these measurements are reported herein.

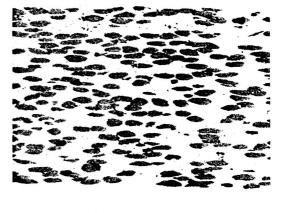


a. 90 W - 10 UO₂ (Photo Supplied by NASA)



b. 80 W - 20 UO₂





d. 60 W - 40 UO₂ (Photo Supplied by NASA)

Figure 1. Photomicrographs of W-UO $_2$ Mixtures

II. MATERIALS

The test materials used in this study were fabricated in the form of plates, about 0.030 in. thick. Powder metallurgy techniques are used to produce sintered compacts of about 60% of theoretical density. Batch sintering these compacts in hydrogen at 3150°F results in a densification to about 92% of theoretical density. The compacts are clad with tungsten, to retain the fuel, and hot rolled to produce a very dense structure. Photomicrographs of samples of W - UO₂ mixtures are shown in Figure 1. The UO₂ is visible as randomly distributed spheroids.

The densities of the samples used in this study were determined by the standard water displacement method. These results are summarized in Appendix I, and are compared to the theoretical values in Figure 2. The densities of the 70-30 and 60-40 mixtures are greater than the theoretical density

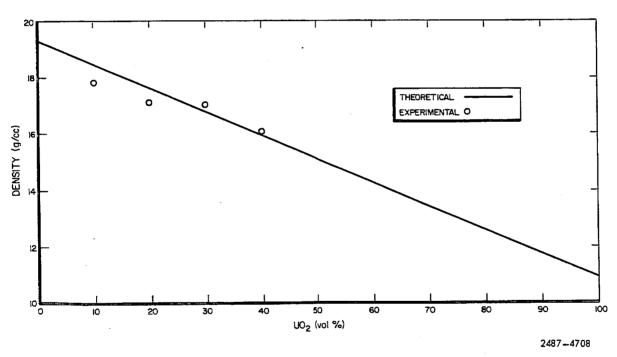
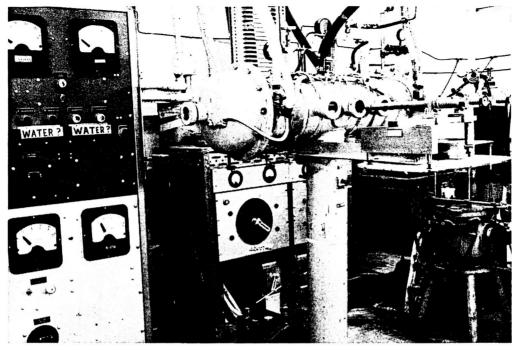


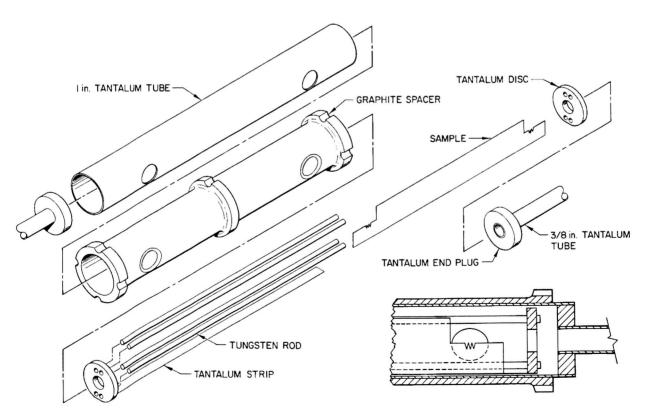
Figure 2. Density of W-UO₂ Mixtures

for W-UO₂ compacts due to the tungsten cladding on the surfaces of the samples. Because of the small sample size used for the 70-30 mixture (<0.2 g, Appendix I), the density obtained for this mixture may not be representative of the bulk material of this composition. Since the density values obtained in this study are close to the theoretical density of the material, the theoretical values were used for conductivity calculations.



7513-1884

a. Photograph



2487 - 4725

b. Specimen Holder

Figure 3. Thermal Expansion Furnace (Figures Unclassified)

III. THERMAL EXPANSION

A. APPARATUS AND EXPERIMENTAL PROCEDURES

The test equipment for measuring linear thermal expansions to very high temperatures is described in detail in the literature. ⁴ Measurements are made by observing the relative displacements of fiducial marks on the opposite ends of 5-in. long specimens as they are heated in a graphite tube furnace. Displacements are measured with two telescopes fitted with filar eyepieces, capable of accurately and reproducibly measuring to ±0.00005 in. Samples are usually allowed to expand freely on a tungsten holder, within a tantalum tube which protects them from contamination. The furnace atmosphere is controllable from vacuum (1 mm Hg) to 200 psi. Often, dried argon at 50 psi is used. Generally, about 2 hr are required to reach a new set temperature and stabilize the furnace. The thermal gradient across the sample varied with temperature, being about 60°F at 1000°F and 20°F at 2000°F.

This apparatus, which is shown in Figure 3, has been used to measure the expansions of molybdenum, tantalum, and graphite to 6500°F. Recently, data on the carbides of silicon, boron, and titanium, as well as the oxides of aluminum, magnesium, and beryllium, were obtained. 6,7

B. RESULTS

The results for the 90-10, 80-20, and 70-30 mixtures are presented in Figure 4. The coefficient of expansion of the mixtures increases with increasing UO₂ content. However, the expansion of all of these mixtures is much closer to that of pure tungsten than it is to an averaged expansion of tungsten and UO₂. In fact, the expansion of the 90-10 mixture is very close to the expansion of pure tungsten. For example, at 2190°F, the expansion of W is 0.60% and that of the 90-10 mixture is 0.605%, while the expansion of UO₂ is 1.27% at this temperature. An average expansion of the 80-20 mixture is 0.74% at 2190°F, while the observed expansion is only 0.64%. The expansion of the 70-30 composition is very close to that of the 80-20 composition.

The least-square equations for the expansion of the 90-10, 80-20, and 70-30 compositions, for $60^{\circ}F < t (^{\circ}F) < 5000^{\circ}F$, are:

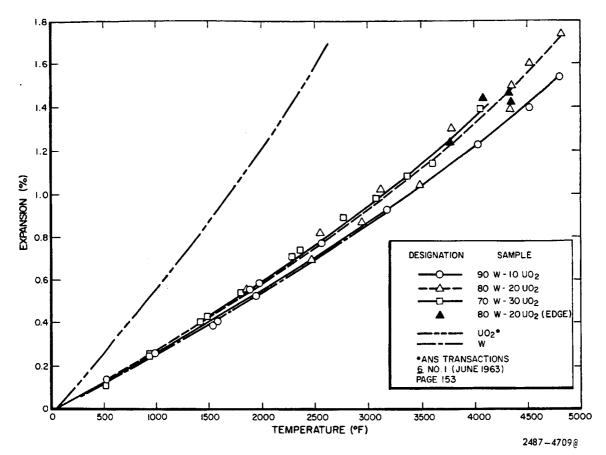


Figure 4. Thermal Expansion of 90 W - 10 UO $_2$, 80 W - 20 UO $_2$, and 70 W - 30 UO $_2$ Mixtures

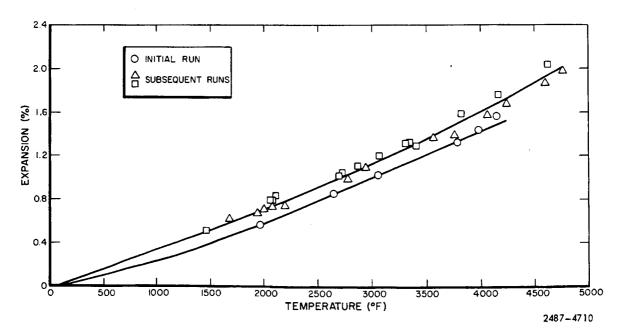


Figure 5. Thermal Expansion of 60 W - 40 UO_2 Mixtures

90-10: % expansion =
$$-1.748 \times 10^{-2} + 2.8997 \times 10^{-4} t - 3.5932 \times 10^{-9} t^2 + 2.1028 \times 10^{-12} t^3$$
 ...(2)
80-20: % expansion = $-2.321 \times 10^{-2} + 3.9013 \times 10^{-4} t - 6.0273 \times 10^{-8} t^2 + 1.1362 \times 10^{-11} t^3$...(3)
70-30: % expansion = $-1.7124 \times 10^{-2} + 2.8387 \times 10^{-4} t + 1.08175 \times 10^{-8} t^2 + 7.4091 \times 10^{-13} t^3$...(4)

These equations are plotted with the experimental data in Figure 4.

By differentiating these equations with respect to t, and accounting for the factor of 100 which changes percent expansion to the decimal, one obtains the equations which express the coefficient of expansion as a function of temperature for each of these compositions:

90-10: coeff. of expansion =
$$2.8997 \times 10^{-6} - 7.1864 \times 10^{-11} t + 6.3084 \times 10^{-14} t^2$$
 ...(5)
80-20: coeff. of expansion = $3.9013 \times 10^{-6} - 1.20546 \times 10^{-9} t + 3.4086 \times 10^{-13} t^2$...(6)

70-30: coeff. of expansion =
$$2.8387 \times 10^{-6} + 2.16350 \times 10^{-10} t + 2.22273 \times 10^{-14} t^2$$
 ...(7)

For example, the coefficient of expansion of the 90-10 mixture at 2000°F is 3.01×10^{-6} /°F.

The initial expansion of one of the samples of the 60-40 mixture, (circles on Figure 5) was very close to that of the 80-20 and 70-30 compositions. However, the thermal expansion measured during subsequent runs on the same specimen

(triangles, Figure 5) and for all runs on a second sample (squares, Figure 5) were appreciably above the initial values. The reason for this behavior is not understood, but may be due to stress-relief during the initial heating to 4200°F. The equation for the expansion, measured during all but the one initial run, is

60-40: % expansion =
$$-2.3323 \times 10^{-2} + 3.8906 \times 10^{-4} t - 1.4755 \times 10^{-8} t^2 + 4.8042 \times 10^{-12} t^3$$
 ...(8)

The expansion of the 80-20 material was measured in the width direction at high temperatures. Due to the sample dimensions, it was not possible to measure the expansion below 3700°F with sufficient accuracy. However, the data above that temperature show that the expansion is isotropic. These data points are included on Figure 4.

IV. SPECIFIC HEAT

A. CALCULATED VALUES

Specific heat values of the various composites were obtained by suitable averaging of the specific heats of the two components (Knoop's rule). Fortunately, the specific heats of tungsten and UO₂ are known to within a few percent. ^{10,11} Consequently, the calculated specific heats should be accurate to within ±4%. The results are plotted in Figure 6.

B. ENTHALPY

The most common technique for determining the specific heat is to measure the increase in enthalpy as a function of temperature, by means of a drop calorimeter. The specific heat, which is the slope of the enthalpy curve, may then be determined graphically or by differentiating the equation which expresses the enthalpy data. In order to check the validity of the specific heat calculations, the reverse procedure was followed (i.e., the equation which expresses the specific heat of the 80-20 mixture was obtained by a computer, and was integrated to obtain enthalpy values which were checked against experimental enthalpy values). The results are plotted in Figure 7. The enthalpy was experimentally measured by Thermatest Laboratories. Three of the enthalpy points had large errors associated with them, because the enthalpy of the graphite crucible used to hold the sample constituted an appreciable portion of the total measured enthalpy. The other points, which were obtained using a copper crucible, were subject to much less error, because of the smaller enthalpy of the copper crucible. When the normal measuring errors, which are estimated at ±5%, are added to the uncertainties caused by the enthalpy of the crucible, the enthalpy values overlap the calculated values above 800°F. Consequently, the calculated specific heat results were used in the subsequent thermal conductivity calculations.

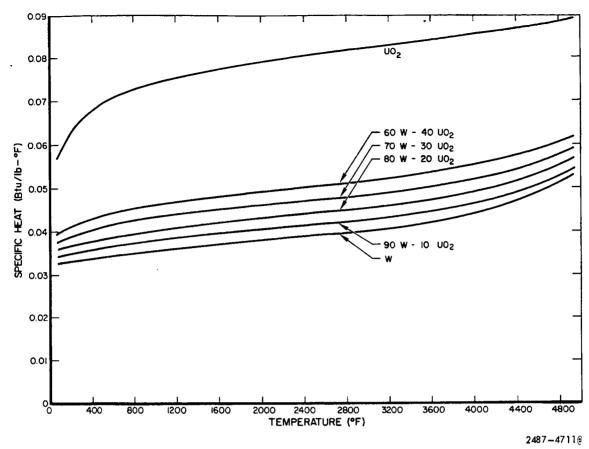


Figure 6. Specific Heat of $W-UO_2$ Mixtures

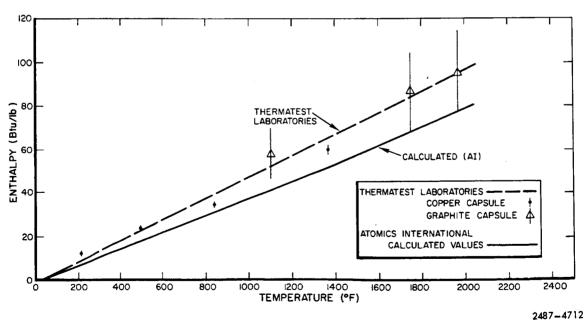


Figure 7. Enthalpy of 80 W - 20 UO_2 Mixtures

V. THERMAL DIFFUSIVITY

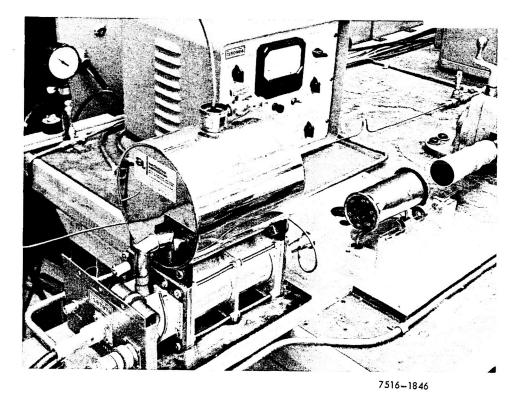
A. APPARATUS

The flash diffusivity method, which was originally described by Parker et al., 12 was used for the thermal diffusivity measurements. In this method, a pulse of energy is radiated to the front face of a disk-shaped sample, and the resulting temperature history of the rear face is used to determine a diffusivity value. The ambient temperature of the sample is controlled by a tantalum tube heater, and the pulse of energy raises the average sample temperature only a few degrees.

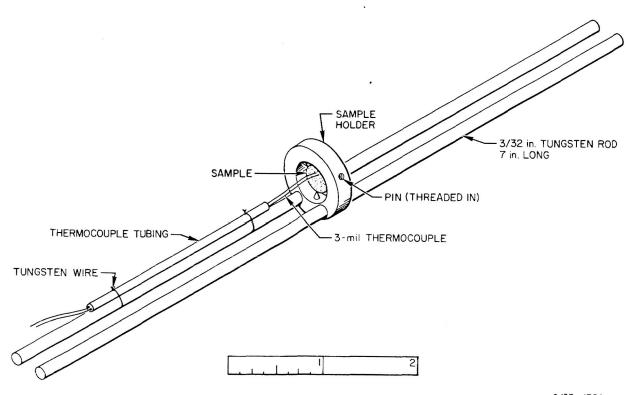
In the original description by Parker et al., 12 a xenon flash lamp was used as the energy source. However, the flash lamp is limited by the fact that the emitting area is so large that optically focusing the energy on a small specimen is not very practical. Therefore, one must keep the flash lamp near the sample, which is not convenient at high sample temperatures. On the other hand, a laser efficiently concentrates the flash lamp energy into a coherent light beam, well suited for irradiating small samples which are enclosed within vacuum furnaces. Consequently, our laboratory uses a laser beam energy source for the flash diffusivity apparatus.

The rear face sample temperature is measured with thermocouples up to 3100°F, and with a photoelectric detector at higher temperatures. Both PbS and photomultiplier tubes have been used as detectors. Since the recording pyrometer has been described in a recent article, ¹³ it will not be discussed here.

The apparatus is shown in Figure 8. A vacuum chamber (not shown) fits over the tantulum tube heater and shield. The mounted sample and radiation shields are shown near the heater. During a measurement, the sample and holder are placed inside the tube heater. The laser is shown in the lower left hand corner of the photograph, and the recording pyrometer is shown lined up with the laser and the tube furnace. The vacuum system is located beneath the table.



a. Photograph



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b. Specimen Holder
Figure 8. Thermal Diffusivity Apparatus
(Figures Unclassified)

B. FINITE PULSE TIME EFFECT

From the transient response of the rear face temperature, the simple relation

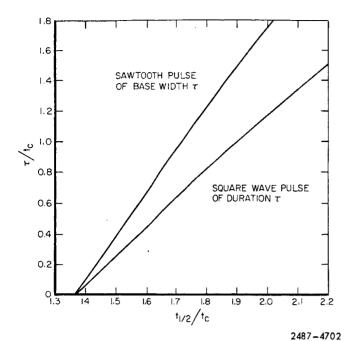
$$\alpha = 1.37 a^2 / \pi^2 t_{1/2}$$
 ...(9)

was obtained by Parker et al., where $t_{1/2}$ is the time required for the backface to reach one-half of its maximum temperature rise, and a is the sample thickness. The derivation of Equation 9 assumes that the energy pulse is received within a time duration (τ), which is short compared to characteristic risetime (t_c) given by

$$t_{c} = a^{2}/\pi^{2}\alpha \qquad \dots (10)$$

Normally, one chooses an appropriate thickness for a given material, such that $t_c > 50 \tau$, and Equation 9 is then obeyed within 2%. However, in the present case, the sample thickness had to be held constant. Since the diffusivity of each mixture is a fundamental property of the material, the characteristic time was fixed for each sample at each temperature. Of course, the characteristic times varied with composition and temperature. However, the range of characteristic times for these materials was from 1 to 7 msec. Consequently, in order to apply Equation 9, τ should be from 0.02 to 0.14 msec. Since the normal duration of the laser burst is from 1.1 to 1.7 msec, this condition is not met. It is possible to shorten the laser pulse by means of "Q" spoilers. This in an expensive and inefficient operation. In fact, about 90% of the available energy is lost, making it difficult to produce a sample temperature rise sufficient for In addition, extremely rapid response thermoaccurate measurements. couples and amplifiers are required for the measurements. Consequently, the mathematical expression which expresses the heat flow was modified to take into account the finite pulse time effect.

The general equation which expresses the heat flow in a thin wafer subjected to an energy pulse on one side was solved by J. Cape and G. Lehman of our laboratory. ¹⁴ This equation takes into account both the finite pulse time effect and radiation losses from the edges. Cape and Lehman solved the radiationless



0.24 0.22 0.20 SAMPLE A (= 0.0622 cm) ■ CORRECTED 0.18 ■ UNCORRECTED SAMPLE B (= 0.0488 cm)
• CORRECTED 0.16 a (cm²/sec) · UNCORRECTED SAMPLE C (ℓ = 0.0784 cm)

♦ CORRECTED 0.14 ♦ UNCORRECTED 0.12 SAMPLE D (= 0.05029 cm)

CORRECTED UNCORRECTED 0.10 0.08 0.06 0.04 600 700 300 400 500 TEMPERATURE (°C) 2487-4704

Figure 9. Finite Pulse Time Corrections (Figure Unclassified)

Figure 10. Thermal Diffusivity of Thin Samples of Armco Iron (Figure Unclassified)

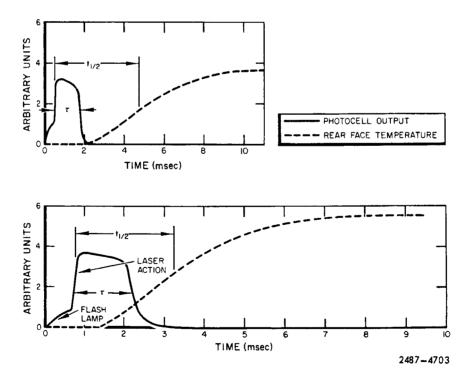


Figure 11. Response of Photocell and Rear Face
Temperature
(Figure Unclassified)

case for a finite sawtooth pulse. However, an examination of the laser pulse revealed that it closely approximated a square wave. Therefore, it was necessary to solve the general expression for a square wave energy input. This is done in Appendix III.

Because there are three variables $(t_c, \tau, and t)$ instead of two, it is not possible to plot $t/t_c \underline{vs} \delta(a,r,t)/\delta(a,r,\infty)$ where $\delta(a,r,t)/\delta(a,r,\infty)$ is the ratio of the rear face temperature rise to the maximum rise, as Parker <u>et al</u>. did when they determined that, when this ratio equals 0.5:

$$t = t_{1/2} = 1.37 t_c$$
 ...(11)

When one substitutes Equation 10 into Equation 11, one obtains Equation 9, which is the expression used to calculate diffusivities from the experimentally measured half-time when τ is small compared to t_c . In the present case, however, it is possible to plot τ/t_c vs $t_{1/2}/t_c$, to yield a curve which can be used to calculate t_c , and hence α , from the experimentally measured values of τ and $t_{1/2}$. In order to obtain such a curve, one must first plot values of $\delta(a,r,t)/\delta(a,r,\infty)$ for selected values of τ/t_c , by varying the values of t/t_c in Equation 22 (Appendix III). From these plots, the value of $t_{1/2}/t_c$ is determined for each of the selected values of τ/t_c .

The graphs of $\tau/t_c \, \underline{vs} \, t_{1/2}/t_c$ for a square wave and sawtooth energy pulse are shown in Figure 9. The curve for the square wave energy input shows, for example, that, when the pulse time is 0.8 of the characteristic time, the numerical factor is 1.785, instead of the 1.37 value used when pulse times are short compared to the characteristic time. For this particular case, then, an error of 30% is involved in neglecting the effect of the finite pulse time.

The validity of the described procedure was verified experimentally using Armco iron, which is the generally accepted thermal conductivity standard. Since the diffusivity of Armco iron decreases rapidly with increasing temperature, the characteristic time has a strong temperature dependency. Of course, the characteristic time can also be changed by varying the sample thickness. Consequently, it is possible to check the mathematics by either measuring the diffusivity of samples of different thickness or by measuring the diffusivity of the same sample at different temperatures. Both methods were used to obtain the results shown in Figure 10. The solid line represents the literature value,

the solid points represent the values obtained from the experimental values of au and $t_{1/2}$, and the unfilled points represent the uncorrected data. These data represent values of τ/t_c ranging from 0.3 to 1.5. Some of the values at room temperature are in error; because the laser pulse is not a true square wave, but lies somewhere between a square wave and a sawtooth function. This is illustrated in Figure 11, which shows the photocell output from the time the flash lamp fires until the output returns to its base line, and also shows the rear face temperature response. For the first 1/2 msec following the initiation of the flash lamp discharge, the laser crystal acts as a "light pipe." Although this energy is apparent to the photocell, it is insignificant as far as heating the sample is concerned, and the pulse width and half-time are measured from the initiation of the laser action, as indicated in the figure. The experimental curves from two separate experiments are shown in Figure 11. The half-time measured from the lower curve can be expected to be influenced to a greater extent by the deviation of the laser pulse from a true square wave, since the rear face temperature is rising during the laser action. Consequently, the true correction for this curve lies between the correction curves for the square and sawtooth energy pulses shown in Figure 9. This causes the corrected data to be several percent high when the half-time is only 1.5 times the pulse width.

C. RESULTS

It was planned originally to use the photoelectric pyrometer for the hightemperature (i.e., > 2500°F) measurements, and thermocouples for the lowtemperature measurements. However, the photoelectric pyrometer went into saturation for 10 msec following the laser pulse. While this does not interfere with obtaining results on samples of 0.1 in. thickness, where the rise time is usually several hundred milliseconds, it prevented the obtaining of data for the thin samples, since their rise time is less than 10 msec. Although this problem could be overcome by modifying the apparatus and obtaining a set of laser interference filters, cost and time elements prevented this being done within the scope of the present contract. Consequently, the measurements were limited to the use of thermocouples. However, tungsten-tungsten/26% rhenium thermocouples were substituted for the Chromel-Alumel thermocouples, in order to extend the practical range to 3800°F. The diffusivity data are so well-behaved that the results can be confidently extrapolated to at least 4500°F, and this was done for the purpose of estimating the thermal conductivity up to this temperature.

The results for two samples of the 90-10 mixture are shown in Figure 12. The diffusivity decreases smoothly from 0.471 cm²/sec at 100°C (212°F) to 0.220 cm²/sec at 2000°C (3632°F). The maximum scatter in the data from the smooth curve is 10%, and the usual scatter is within 5%.

The results for the 80-20 mixture are shown in Figure 13. Two different thicknesses (0.0281 and 0.0605 in.) were used in these experiments. Since the thickness enters into the equation as the square term, the use of these two thicknesses is equivalent to changing the experimental conditions by a factor of four. Consequently, the agreement between the results for the two thicknesses constitutes a verification of the experimental values. The diffusivity of the 80-20 mixture decreases from 0.313 cm²/sec at 100°C to 0.197 cm²/sec at 2000°C.

The diffusivity of the 80-20 mixture was also measured in the length direction, and it was found to be significantly greater than that measured in the width direction. This is shown in Figure 13. Because of the sample geometry, measurements made in the length direction do not require the finite pulse time correction discussed previously. However, due to heat loss from the sides of the specimen, the measurements are limited to low temperatures. Nevertheless, the measurements show that the thermal conductivity, unlike the thermal expansion, is anisotropic, with the thermal conductivity in the length direction being $20 \pm 10\%$ greater than the conductivity in the width direction.

The results for the thermal diffusivity of the 70-30 mixture are given in Figure 14, and the results for the 60-40 mixture are given in Figure 15. The reproducibility of these data is better than that obtained on the 90-10 and 80-20 mixtures, because of the lower diffusivities and consequently longer characteristic times.

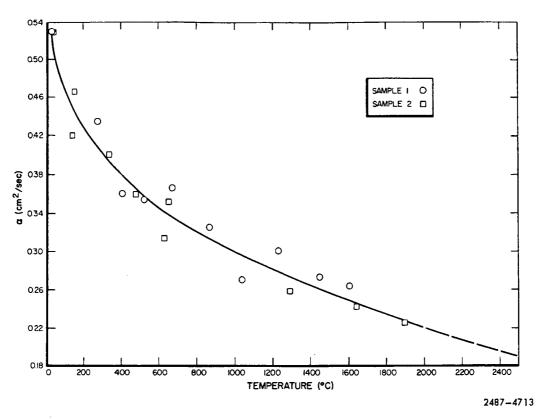


Figure 12. Thermal Diffusivity of 90 W - 10 UO_2 Mixture

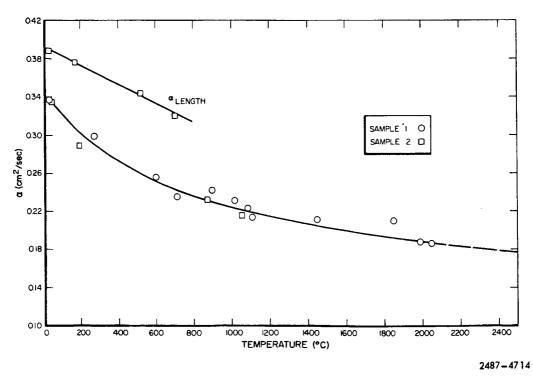


Figure 13. Thermal Diffusivity of 80 W - 20 UO_2 Mixture

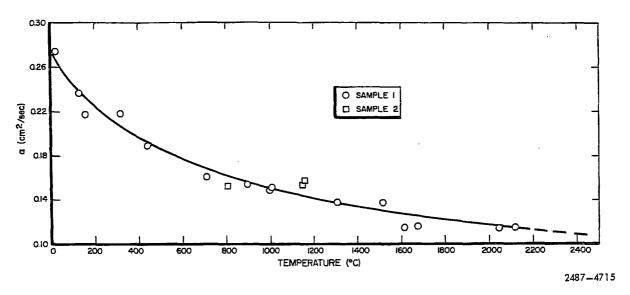


Figure 14. Thermal Diffusivity of 70 W - 30 UO_2 Mixture

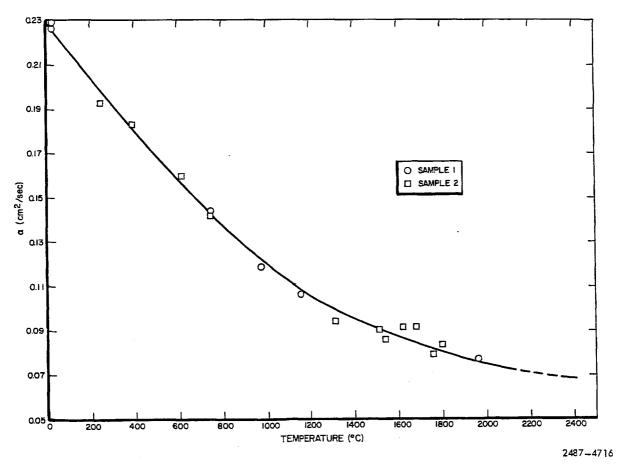


Figure 15. Thermal Diffusivity of 60 W - 40 UO_2 Mixture

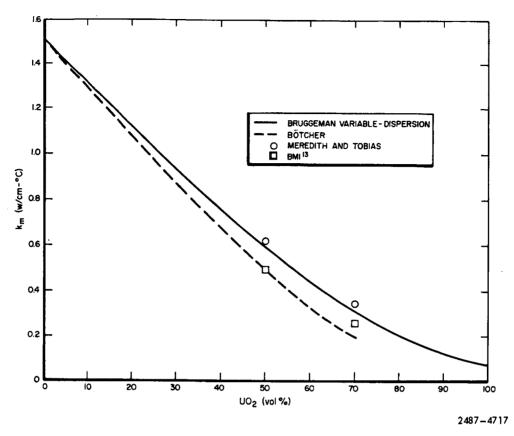


Figure 16. Thermal Conductivity of W-UO $_2$ Mixtures at 200°C

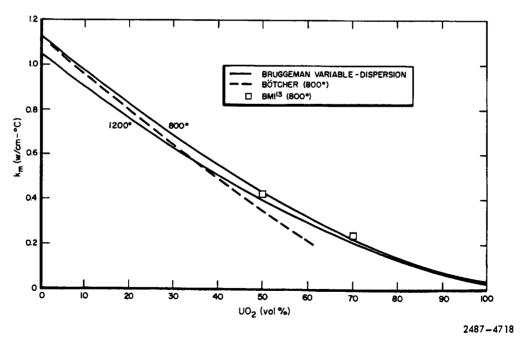


Figure 17. Thermal Conductivity of W-UO $_2$ Mixtures at 800 and 1200° C

VI. THERMAL CONDUCTIVITY

A. VALUES OBTAINED FROM MIXTURE CALCULATIONS

Numerous equations have been derived for the calculation of the thermal conductivities of mixtures, based upon the conductivities of the components. These equations have been summarized by Powers, ¹⁵ who classified the equations according to the basic assumptions concerning the nature of the mixture. Three equations, the Bruggeman variable-dispersion, the Bötcher, and the Meredith and Tobias, were investigated, since the assumptions made in their derivations are believed to be approximately true in the case of the W-UO, dispersions.

The Bruggeman variable-dispersion equation, which assumes that spherical particles are imbedded in a continuous phase, is given by

$$1-P_1 = \frac{K_1 - K_m}{K_1 - K_2} \left(\frac{K_2}{K_m}\right)^{1/3} \qquad \dots (12)$$

where:

P₁ = volume fraction of the discontinuous phase

K₁ = conductivity of the discontinuous phase

 K_2 = conductivity of the continuous phase

 $K_m = conductivity of the mixture.$

For any given temperature, K_1 and K_2 are constant. Consequently, it is possible to plot K_m as a function of P_1 , in order to determine the conductivity of any composition. Figure 16 is such a plot for W-UO₂ at 200°C. Experimental values, obtained by BMI on 50-50 and 30-70 vol% W-UO₂ mixtures, ¹⁶ are also shown in the figure. The calculated values are approximately 20% above the experimental results for both compositions. A similar plot for 800 and 1200°C is shown in Figure 17. From this figure, it can be seen that the experimental results are only a few percent different from the calculated values at 800°C. No experimental values were determined by BMI above 1000°C.

Botcher 15 devised an equation, based on the assumption that each particle is surrounded by a medium of K_m , and not necessarily surrounded completely by the opposite phase. It is:

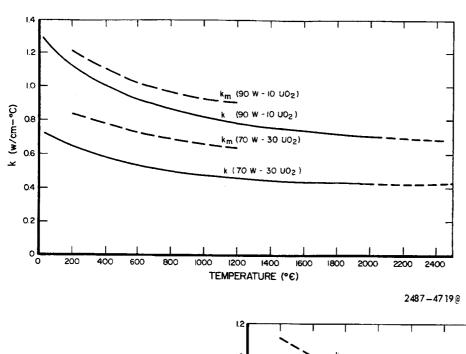
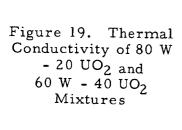
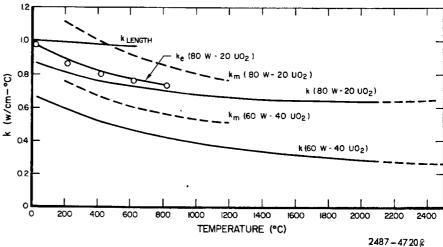


Figure 18. Thermal Conductivity of 90 W - 10 UO₂ and 70 W - 30 UO₂ Mixtures





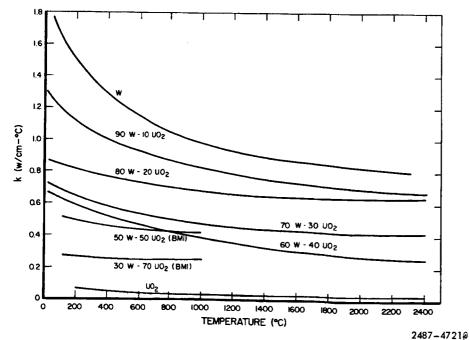


Figure 20. Thermal Conductivity of W-UO₂
Composites

$$\frac{K_{m} - K_{1}}{3K_{m}} = P_{2} \frac{K_{2} - K_{1}}{K_{2} + 2K_{m}}.$$
 (13)

The results for W-UO₂ at 200°C are given in Figure 16, and the results at 800°C are included in Figure 17. It should be noted that the agreement between the experimental results and the calculated values is better at 200°C for the Bötcher equation, and better at 800°C for the Bruggeman variable-dispersion equation.

Recently, Meredith and Tobias, ¹⁵ using the Maxwell equation, derived a dispersion equation for variable concentrations, based on the assumption of only two particle sizes, rather than a large number of particle sizes, as assumed in the Bruggeman variable-dispersion equation. Their equation is:

$$K_{m} = K_{2} \left[\frac{4K_{2} + 2K_{1} - 2P_{1}(K_{2} - K_{1})}{4K_{2} + 2K_{1} + P_{2}(K_{2} - K_{1})} \right] \left[\frac{4K_{2} + 2K_{1} - P_{2}(4K_{2} - K_{1})}{4K_{2} + 2K_{1} - P_{2}(K_{2} + 2K_{1})} \right] (14)$$

As shown in Figure 16, the results of this equation are very close to that of the Bruggeman variable-dispersion equation.

The experimental results for the present samples are compared to the values calculated, using the Bruggeman variable-dispersion equation in the following sections.

B. VALUES OBTAINED FROM THERMAL DIFFUSIVITY

Equation 1 was used to compute the thermal conductivities of the various mixtures. These calculations are summarized in Appendix V.

The results for the 90-10 and 70-30 mixtures are plotted in Figure 18, and the results for the 80-20 and 60-40 mixtures are plotted in Figure 19. The conductivities of each of the mixtures decreases with increasing temperature; and, as noted in Section V, the conductivity is anisotropic.

The calculated values for k_m are also plotted in Figure 18 and 19. From these figures, it can be seen that k_m is greater than the experimentally observed k. This is due, at least in part, to the anisotropic effect, and is in semiquantitative agreement with predictions based on Power's report.

The experimental thermal conductivity results, for the various composites and for pure W and UO₂, are summarized in Figure 20.

C. VALUES OBTAINED FROM ELECTRICAL RESISTIVITY

Contemporary theory states that the heat conduction in electrically conducting materials is the sum of the electronic conduction (k_e) and phonon (lattice) conduction (k_p) , i.e.,

$$k_t = k_e + k_p$$
 , ...(15)

where k_t is the total heat conductivity. The heat conduction due to electrons is given by the Wiedemann-Franz relation

$$k_e = \frac{LT}{\rho} \qquad \dots (16)$$

where:

L = a constant, theoretically equal to $2.45 \times 10^{-8} \text{ w-}\Omega/\text{°K}^2$

T = the absolute temperature (°K)

 ρ = the electrical resistivity, (Ω -cm)

Although the phonon conduction should follow a 1/T law, there is ample evidence that it may not do this. ¹⁷ In any event, the total conductivity must be equal to or greater than the electronic conductivity alone.

The electrical resistivity of the 80-20 material was measured by the dc potentiometric drop method, in which the voltage drop caused by the passage of a dc current through the sample is compared to the voltage drop across a standard resistor. The current is then reversed and the process repeated, in order to eliminate the effects of stray emfs.

The results, along with literature values for tungsten, are given in Figure 21. These values were used to calculate the thermal conductivity contributon due to electrons, and the results are plotted in Figure 19. Since the electrical resistivity was measured in the length direction, k_e should be compared to the k_e obtained for the length direction. When this is done, one notes that k_e is less

than k, as it should be, and that about 80% of the heat is transported by electrons at $400\,^{\circ}$ C. This is in good agreement with the results from BMI. 16

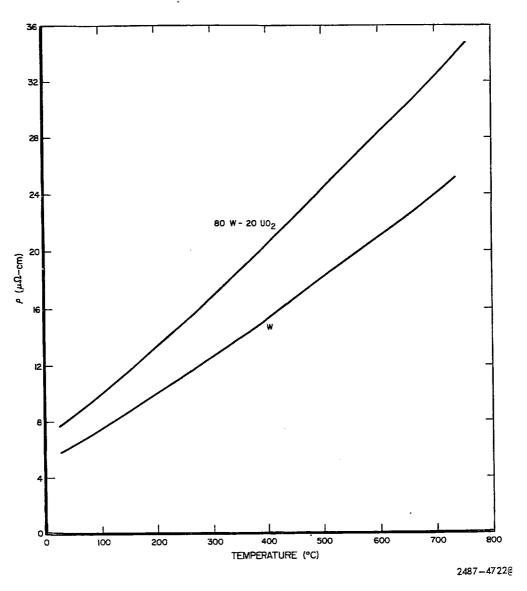


Figure 21. Electrical Resistivity of 80 W - 20 UO_2 Mixture

REFERENCES

- 1. F. E. Rom, P. G. Johnson, and R. E. Hyland, "Water-Moderated Nuclear Rockets," NASA TM-X482 (1961)
- 2. R. J. Buzzard and F. L. Gill, "High Temperature Mechanical Properties of a Tungsten-Uranium Dioxide Composite," NASA TM (E-2453) (1964)
- 3. N. T. Saunders, G. K. Watson, and R. J. Buzzard, "Investigation of Tungsten and Uranium Dioxide," Proceedings of Nuclear Propulsion Conference, August 15-19, 1962, Naval Postgraduate School, Monterey, California, TID-7553, Pt. II, Book 2
- 4. N. S. Rasor and J. D. McClelland, "Thermal Property Measurements at Very High Temperatures," Rev. Sci. Inst. 31, 595 (1960)
- 5. N. S. Rasor and J. D. McClelland, "Thermal Properties of Graphite, Molybdenum, and Tantalum to Their Destruction Temperatures," Int. J. Phys. Chem. of Solids 15, 17 (1960)
- 6. R. E. Taylor, "Thermal Conductivity and Expansion of Beryllia at High Temperatures," J. Am. Ceram. Soc. <u>45</u>, 74 (1962)
- 7. C. J. Enberg and E. H. Zehms, "Thermal Expansion of Al₂O₃, BeO, MgO, B₄C, SiC, and TiC above 1000°C," J. Am. Ceram. Soc. <u>42</u>, 300 (1959)
- 8. F. A. Mauer and L. H. Bolz, "Measurement of Thermal Expansion of Cermet Components of High Temperature X-ray Diffraction," WADC-TR-55-473 (December 1955)
- 9. J. Belle, Uranium Dioxide: Properties and Nuclear Applications, Naval Reactor, Division of Reactor Development, United States Atomic Energy Commission (1961)
- 10. V. A. Kirillin, A. E. Sheindlin, and V. Y. Chekhovskoy, "Enthalpy and Heat Capacity of Some Solid Materials at Extremely High Temperatures," Revised Supplement to Advance Papers, International Symposium on High Temperature Technology (September 8-11, 1963) Stanford Research Institute pp 110-133. Proceedings to be published by the International Union for Pure and Applied Chemistry.
- 11. H. L. Schick et al., "Thermodynamics of Certain Refractory Compounds," Part II, Fourth Quarterly Progress Report, RAD-SR-63-105 (June 1963)
- 12. W. J. Parker, R. J. Jenkins, C.P. Butler, and G. S. Abbott, "Flash Method of Determining Thermal Diffusivity, Heat Capacity, and Thermal Conductivity," J. App. Phys. 32, 1679 (1961)
- 13. R. E. Taylor and M. M. Nakata, "Thermal Properties of Refractory Materials," WADD-TR-60-581 Pt IV (November 1963)

REFERENCES

- 14. J. Cape and G. W. Lehman, "Temperature and Finite Pulse-Time Effects in the Flash Method for Measuring Thermal Diffusivity," J. Appl. Phys. 34, pp 1909-1913 (July 1963)
- 15. A. E. Powers, "Conductivity in Aggregates," KAPL-2145 (1961)
- 16. R. W. Dayton and R. F. Dickerson, "Progress Relating to Civilian Applications During February 1963," BMI-1619 (March 1963)
- 17. R. E. Taylor and J. Morreale, "Thermal Conductivity of Titanium Carbide, Zirconium Carbide, and Titanium Nitride at High Temperatures," J. Am. Ceram. Soc. <u>47</u>, pp 69-73 (February 1964)

APPENDIX I SAMPLE DENSITIES

		Sample Composition														
	90 W - 10 UO ₂	80 W - 20 UO ₂	70 W - 30 UO ₂	60 W - 40 UO ₂												
Mass in air (g)	2.1494	1.1880	0.1979	1.3529												
Mass in water (g)	2.0296	1.1192	0.1863	1.2692												
Temperature of water (°C)	22.1	22.1	21.9	21.9												
Volume of sample (cc)	0.1204	0.0692	0.0116	0.0841												
Density of sample (g/cc)	17.85	17.17	17.06	16.09												

APPENDIX II THERMAL EXPANSION OF W-UO₂ MIXTURES

%E	0.75	သံဖ	∞	۲.	∞	0.	0	0.	~;	_	0	1.20	٠,	ς,	κĵ	ς,	٤,	ς,	5.	4.	1.58	ιú	7.	9	∞	0.	6.
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ာ့	, , ,	7	15	20	45	1488	50	53	57	62	98	69	8 1	84	87	96	07	60	1	19	24	28	30	34	53	55	63
Sample	- M 09	$40~\mathrm{UO}_2$	(Contin-	ued)																							
%E	1.24	4.	4.	4.		┪.	7	7	4.	4.	ι.	7	7.	∞	6.	0.	1.14	3				ນຸ	ð	9	ď	0.72	∞
유.	3763	80	33	34		2	4	S	42	49	80	29	37	77	08	36	3612	90				46	69	94	67	2012	2
ာ့	07	2250	39	39		2	0	\vdash	~	$\overline{}$	∞	25	29	52	69	85	1989	23				6	~ 2	90	08	0	1133
Sample	80 W -	20 UO2	(Edge)	0		70 W -	30 UO	,														- M 09	40 UO	7			
%E	Τ.	7	~	3	4	0.56	7	Ŋ	9	7	6	. 2	4	'n			0.70	α	α		0	γ,	\ \ !			1.60	1.74
ە. ئىتا	1 2	∞	6	44	49	1886	94	99	, 4 , 0	5.7	<u>~</u>	03	52	80)	23	47	. r.	95	12	49	79	66	7,	ر ب بر	, rc	4827
ာ့						1030	90	90	36,		7.	5	49	65))	0	ر 1 تر	4.0	6.2	, <u>-</u>	9.2	į	200	3 6	, 4	49	2664
Sample	- M 06	10 110) }													80 W =	20 110	,									

APPENDIX III FINITE PULSE TIME EFFECT

Using the notation of Cape and Lehman's paper, 14 the heat flow can be expressed as

$$(k/q_0)\delta(a,r,t) = D_0(r,y_r) \sum_{m=0}^{\infty} C_m X_m \int_0^t dt' W(t') \exp \left[\omega_{om}(t-t')/t_c\right] \dots (17)$$

For a square wave input,

$$W(t') \approx \tau$$
, $t \leq \tau$
= 0, $t > \tau$

For a radiationless case,

$$C_{m}X_{m} = 2\left(\frac{\alpha}{a}\right)(-1)^{m}$$
, $m = 1,2,3...$
 $C_{o}X_{o} = \frac{\alpha}{a}$
 $\omega_{oo} = o$
 $\omega_{om} = -m^{2}$, $m = 1,2,3...$

for $t \le \tau$,

$$(k/q_{o})\delta(a,r,t) = D_{o}(r,y_{r})\frac{\alpha}{a} \left\{ \int_{0}^{t} dt' \frac{1}{\tau} + \frac{2}{\tau} \sum_{m=1}^{\infty} \int_{0}^{t} dt' \exp \left[-m^{2}(t-t')/t_{c}\right] \right\}$$

$$= D_{o}(r,y_{r})\frac{\alpha}{a} \left\{ \frac{t}{\tau} + 2\frac{t_{c}}{\tau} \sum_{m=1}^{\infty} \frac{(-1)^{m}}{m^{2}} \left[1 - \exp\left(-m^{2}t/t_{c}\right)\right] \right\} . \qquad ... (18)$$

for $t > \tau$,

$$(k/q_0)\delta(a,r,t) = D_0(r,y_r)\frac{\alpha}{a} \left\{ \int_0^{\tau} dt' \frac{1}{\tau} + \frac{2}{\tau} \sum_{m=1}^{\infty} (-1)^m \int_0^{\tau} dt' \exp\left[-m^2(t-t')/t_c\right] \right\}$$

$$= D_0(r,y_r)\frac{\alpha}{a} \left\{ 1 + \frac{2t_c}{\tau} \sum_{m=1}^{\infty} \frac{(-1)^m}{m^2} \exp\left(-m^2t/t_c\right) \left[\exp\left(m^2\tau/t_c\right) - 1\right] \right\} \dots (19)$$

But, for $t = \infty$,

$$k/q_0 \delta(a,r,\infty) = D_0(r,y_r) \frac{\alpha}{a}$$
 . (20)

Therefore,

$$\frac{\delta(a,r,t)}{\delta(a,r,\infty)} = \frac{t}{\tau} + 2\frac{t}{\tau} \sum_{m=1}^{\infty} \frac{(-1)^m}{m^2} [1 - \exp(-m^2 t/t_c)], \quad t \le \tau \quad ... (21)$$

$$\frac{\delta(a,r,t)}{\delta(a,r,\infty)} = 1 + 2\frac{t_c}{\tau} \sum_{m=1}^{\infty} \frac{(-1)^m}{m^2} \exp(-m^2 t/t_c) [\exp(m^2 \tau/t_c) - 1] \quad t \ge \tau \quad ... (22)$$

APPENDIX IV

THERMAL DIFFUSIVITIES OF W-UO2 MIXTURES

Sample	Temperature (°C)	Diffusivity (cm ² /sec)	Sample	Temperature (°C)	Diffusivity (cm ² /sec)
90 W - 10 UO ₂	25	0.530	70 W - 30	708	0.160
7	272	0.434	UO ₂	895	0.154
	521	0.354		266	0.148
	677	0.369		1305	0.150
	3070	0.301		חבים ב	0.13
	1039	0.320		1101	0.137
	1230	0.2.0	_	1671	0.116
	1459	0.274		1 201	0.110
	1602	0.265	~	392	168
))	804	0.153
2	477	0.360		1157	0.157
	628	0.315		1147	0.153
	650	0.353		2050	0.155
	336	0.404		2115	0.115
	152	0.465		25	0.275
	30	0.530		128	0.237
	136	0.450		152	0.217
	1298	0.259		311	0.219
	1644	0.242		437	0.189
	1900	0.225			
		,	60 W - 40	52	0.229
80 W - 20 UO,	25	0.338	no	744	0.144
1	270	0.298	7	1152	0.107
	595	0.255	7	970	0.119
	402	0.235			
	006	0.243	2	25	0.227
	1022	0.231		239	0.193
	1102	0.214		393	0.182
	1088	0.223		615	0.160
	1.450	0.212		745	0.142
	1850	0.220		1315	0.0946
	3050	0.107		1541	0.0808
	0607	0.105		1615	0.0925
•	ű	000		1253	0.0925
٧	176	0.230		1510	0.000
	898	0.233		0621	0.0843
	1052	0.214		1958	0.0778
80 W - 20 UO.	25	0.389			
1 2	170	0.377			
(Length Direction)	513	0.344			
	200	0.320			

APPENDIX V THERMAL CONDUCTIVITIES OF W-UO2 COMPOSITES

																																							Ī
k (w/cm-°C)		0.721	0.638	0.582	0.544	0.511	0.482	777	0.4.0	0.447	0.434	0.425	0.417	0.419	0.420				0.601	0.592	0.542	0.487	7.70	0000	0.388	0.348	0.322	0.305	0.283	0.269	0.264	0.265							
(g/cc)	ادہ	17.06	17.06	17 06	17.06	17.06	17.06	17.00	90.71	17.06	17.06	17.06	17.06	17.06	17.06		re		16.09	16.09	16.09	16.09	10.07		9	9			9	9	16.09								
(cm ² /sec)	30 UO2 Mixtur	2	0.220	_	_	-	٠,	┙,	_	_	_	_			. –	7	40 UO, Mixtur		0.226	0.202	0 178	21.0	0.155	0.135	0.118	0.104	0.0940	0.0870	0.0800	0.0742	0.0705	0.0680							
(cal/g-°C)	70 W - 3	0.0377	0.0406	0.0422	20.0	0.0400	0.044	0.0456	0.0465	0.0474	0.0482	0.0492	0.0503	0.620	0.0320	0.0343	- M 09		0.0395	0.0223	2270	0.0432	0.0467	0.0478	0.0488	0.0497	0.50.0	0.000	0.0525	0.00	0.00	0.0223			_				
Temperature (°K)		506		000	000	006	1100	1300	1500	1700	0001	2100	0000	0057	2500	2700			300	000	200	007	006	1100	1300	0041		1000	1900	0017	7500	0007	00/7			-			
k (w/cm-°C)			1.307	1.065	0.983	0.917	0.872	0.832		0.000	27.0	0.743	0.715	0.690	6.679	0.678				∞	~	5	-	- r	- \	0 \	9	Ψ,	w.	Ä	0.638	۳.	۳.				1.02	0.994	201.0
d (g/cc)	1 0		17.85	7.8	7.8	∞	7		9 (∞	æ.	8.	œ	₹.	7.8	7.8		e e	i_	17.17	17.17	17.17	17 17		17.17	17.17	17.17	17.17	17.17	17.17	17.17	17.17	17.17		hirection	12	- ~	17.17	_
α (cm ² /sec)	10 IIO2 Mixture		0.530	0.423	0.376	0.343	0.0	0.510	0.297	0.279	0.263	0.248	0.234	0.220	20	0,196		20 UO ₂ Mixtur	3	0.337	0.288	0.22.0	3.0	0.250	0.235	0.222	0.213	0.206	0.198	0.193	0.188	0.183	0.179		in the Width		0.388	0.350	0.330
cp (C)		=	0.0330	0.0340	0.0350	0.0350	00000	0.0367	0.0375	0.0385	0.0393	0.0401	0.0409	0.0420	0.0437	0.0463		80 W - 2		0 0360	0.0385	2000.0	0.0395	0.0406	0.0417	0.0426	0 0435	0.0445	0.0452	0.0461	0.0472	0.0400	0.0515	1	W - 20 UO ₂	:	0.0360	0.0395	0.0406
Temperature	(W)		300	000	0 0	007	006	1100	1300	1500	0021	1900	2,00	0017	2300	2200	20			000	000	200	200	006	1100	1300	0 0	1200	1000	1,00	2300	7200	2200	7	80		300	2002	006

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